

Program 1 : Pairing non-diagonal indices of stiffness matrix

```
Pair_STF (B#) := | nr# := rows (B#) - 1
                  for k ∈ [1..(nr#)]
                  | Z#_k := stack (B#_k, B#_k+1)
                  Z#
```

appVersion(4) = "1.0.8348.30405"

appVersion(-4) = "1.0.8348.30405"

Define

```
S_BEAM := [ 4 2
            2 4 ]
```

Program 2 : To find the Stiffness Matrix : Calls Program 1
fdn#=1 if foundation column base is FIXED ", fdn#=0.75 if foundation column base is PINNED

```
Find_K (L#, I#_UC, I#_LC, I#_beam, H#_UC, H#_LC, fdn#) := | nr := rows (L#) + 1
                                                           for j ∈ [1..(nr)]
                                                           |
                                                           |
                                                           | 
$$M#_j := 4 \cdot E \cdot \begin{cases} \frac{I#_{UC_j}}{H#_{UC_j}} + \frac{I#_{LC_j} \cdot fdn#}{H#_{LC_j}} + \frac{I#_{beam_j}}{L#_j} & \text{if } j = 1 \\ \frac{I#_{UC_j}}{H#_{UC_j}} + \frac{I#_{LC_j} \cdot fdn#}{H#_{LC_j}} + \frac{I#_{beam_j}}{L#_{j-1}} + \frac{I#_{beam_j}}{L#_j} & \text{if } 1 < j < nr \\ \frac{I#_{UC_j}}{H#_{UC_j}} + \frac{I#_{LC_j} \cdot fdn#}{H#_{LC_j}} + \frac{I#_{beam_{j-1}}}{L#_{j-1}} & \text{otherwise} \end{cases}$$

                                                           |
                                                           |
                                                           | (STF := Diag (M#)) = "Create diagonal matrix"
                                                           | C# := [1..nr]
                                                           | (D# := Pair_STF (C#)) = "Indices of non-diagonal elements"
                                                           | Clear(j)
                                                           | for j ∈ [1..rows (L#)]
                                                           | | 
$$\left( M2#_j := 2 \cdot E \cdot \frac{I#_{beam_j}}{L#_j} \right) = \text{"Non-diagonal elements"}$$

                                                           | | 
$$\left( r_j := D#_{j_1} \right) = \text{"Row index of non-diagonal elements"}$$

                                                           | | 
$$\left( c_j := D#_{j_2} \right) = \text{"Column index of non-diagonal elements"}$$

                                                           | | 
$$\left( STF_{r_j c_j} := M2#_j \right) = \text{"Assign appropriate M2# to STF"}$$

                                                           | | 
$$\left( STF_{c_j r_j} := M2#_j \right) = \text{"Assign appropriate M2# to STF"}$$

                                                           | STF
```

PROGRAM 3 : To find joint moments & ST

```
Find_ST# (Z, aa (■), kk) := | j := [1..rows (Z)]
                             | jm_j := aa (Z_j)
                             | st_j := -kk-1 · jm_j
                             | st
```

Program 4 : Any function can be passed to this program

```
Arrange (Z#, a (■)) := | j := [1..rows (Z#)]
                       | D_j := a (Z#_j)
```

Program 5 : Find lower column moments. Valid even if foundation columns have equal or diff heights

```
Find_MLC (st, d) := | for j ∈ [1..rows (st)]
                     | a# := st_j
                     | A#_j := a# · d
                     | A#
```

Program 6 : Calls Program 4

```
Pair (B#) := | nrr := rows (B#)1 - 1
               | for k ∈ [1..(nrr)]
               | | Z#_k := [ Y Y_k
                           Y Y_k+1 ]
               | FF (YY) := Z#
               | Arrange (B#, FF (YY))
```

Program 7 : Main program to find support moments, upper & lower column moments
Calls Programs 3, 4, 5 & 6

```
NEW_BMM (ff#, K#, h#UC, h#LC, Ib, Ic, IL, fdn#, L#) :=
  for j ∈ [1..cols(ff#)]
    fem#j :=  $\frac{1}{12} \cdot \left( \overrightarrow{ff\#}_{1j} \cdot L\#^2 \right)$ 
    nrr := rows(fem#)1 + 1
    for j ∈ [1..nrr]
      M#j :=  $\begin{cases} -Y_j & \text{if } j = 1 \\ Y_{j-1} - Y_j & \text{if } 1 < j < nrr \\ Y_{nrr-1} & \text{otherwise} \end{cases}$ 
      F(Y) := M#
      st := Find_ST#(fem#, F(Y), K#)
      stt := Pair(st)
      for j ∈ [1..cols(ff#)]
        for k ∈ [1..rows(L)]
          Msupj k :=  $\left( \frac{stt_{jk} \cdot E \cdot I_{b k}}{L_k} \right)^T \cdot S_{BEAM} + \begin{bmatrix} -fem\#_{jk} \\ fem\#_{jk} \end{bmatrix}^T$ 
          Col_bott# :=  $\frac{I_L}{h\#_{LC}} \cdot 4 \cdot E \cdot fdn\#$ 
          MLCC := Find_MLC(st, Col_bott#)
          Col_upp# :=  $\frac{I_C}{h\#_{UC}} \cdot 4 \cdot E$ 
          MUCC := Find_MLC(st, Col_upp#)
          augment(Msup, MUCC, MLCC)
```

Define

kNm := kN m

Assume

E := 1 MPa

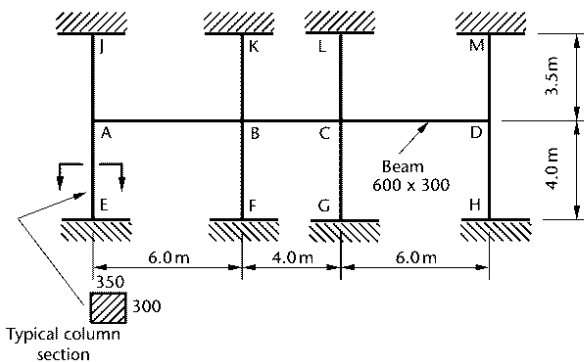
Assume Young's Modulus for RC: This is not really required, but used only for consistent units in intermediate calculations. We can use any numerical value. Ex E = 1 MPa

Ex 3.3 - Page 44 : Reinforced concrete design to Eurocode 2 : *Mosely, Bungey & Hulse - 7th Ed - pal grave macmillan*

EXAMPLE 3.3

Analysis of a substitute frame

The substitute frame shown in figure 3.12 is part of the complete frame in figure 3.10. The characteristic actions carried by the beams are permanent actions (including self-weight) $G_k = 25 \text{ kN/m}$, and variable action, $Q_k = 10 \text{ kN/m}$, uniformly distributed along the beam. The analysis of the subframe will be carried out by moment distribution: thus the member stiffnesses and their relevant distribution factors are first required.



Member Stiffness (ST)

cond = 0.75 when Pinned at base.

cond = 1 when fixed at both ends.

This program can handle both cases as shown below.

cond := $\begin{bmatrix} \text{"Pinned"} & 0.75 \\ \text{"Fixed"} & 1 \end{bmatrix}$



Pinned



Fixed

FDN

DATA

4 upper columns, 4 lower columns and 3 beams

Foundation Columns Fixed at Base

$F_{DN} = 1$

Column

$$b_{UC} := \begin{bmatrix} 300 \\ 300 \\ 300 \\ 300 \end{bmatrix} \text{ mm}$$

$$h_{UC} := \begin{bmatrix} 350 \\ 350 \\ 350 \\ 350 \end{bmatrix} \text{ mm}$$

$$b_{LC} := \begin{bmatrix} 300 \\ 300 \\ 300 \\ 300 \end{bmatrix} \text{ mm}$$

$$h_{LC} := \begin{bmatrix} 350 \\ 350 \\ 350 \\ 350 \end{bmatrix} \text{ mm}$$

$$H_{UC} := \begin{bmatrix} 3.5 \\ 3.5 \\ 3.5 \\ 3.5 \end{bmatrix} \text{ m}$$

$$H_{LC} := \begin{bmatrix} 4 \\ 4 \\ 4 \\ 4 \end{bmatrix} \text{ m}$$

Beam

$$b_{beam} := \begin{bmatrix} 300 \\ 300 \\ 300 \end{bmatrix} \text{ mm}$$

$$h_{beam} := \begin{bmatrix} 600 \\ 600 \\ 600 \end{bmatrix} \text{ mm}$$

$$L := \begin{bmatrix} 6 \\ 4 \\ 6 \end{bmatrix} \text{ m}$$

Define

$$kNm := \text{kN m}$$

Assume

$$E := 1 \text{ MPa}$$

Flanged Beams can be effective only in case of sagging as the concrete is able to take compression. In limit state design we neglect the tensile strength of concrete. As a result, at supports, continuous 'T' beams should be designed as a **rectangular beam** (with width equal to the web width).

Define

$$Line_{zero} := \begin{bmatrix} 0 & 0 \\ \sum \left(\frac{L}{m} \right) & 0 \end{bmatrix}$$

Define

$$LL [1..rows(L)] := 0 = \begin{bmatrix} 0 \\ 0 \\ 0 \\ 0 \end{bmatrix}$$

Moment of Inertia

Upper Column

$$I_{UC} := \frac{1}{12} \cdot b_{UC} \cdot (h_{UC})^3 = \begin{bmatrix} 1.0719 \cdot 10^9 \\ 1.0719 \cdot 10^9 \\ 1.0719 \cdot 10^9 \\ 1.0719 \cdot 10^9 \end{bmatrix} \text{ mm}^4$$

Lower Column

$$I_{LC} := \frac{1}{12} \cdot b_{LC} \cdot (h_{LC})^3 = \begin{bmatrix} 1.0719 \cdot 10^9 \\ 1.0719 \cdot 10^9 \\ 1.0719 \cdot 10^9 \\ 1.0719 \cdot 10^9 \end{bmatrix} \text{ mm}^4$$

Beams

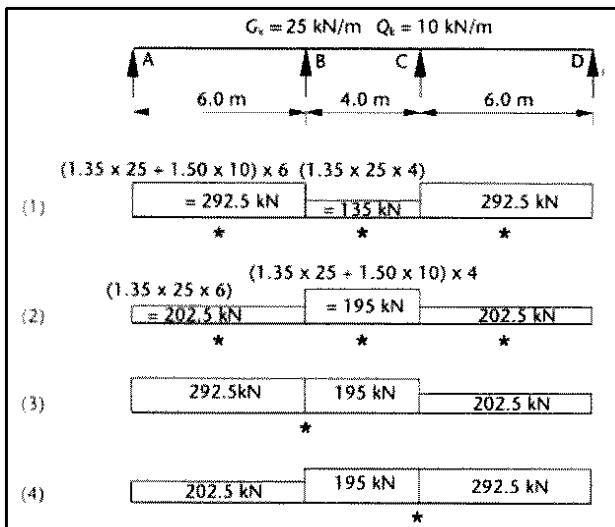
$$I_{beam} := \frac{1}{12} \cdot b_{beam} \cdot (h_{beam})^3 = \begin{bmatrix} 5.4 \cdot 10^9 \\ 5.4 \cdot 10^9 \\ 5.4 \cdot 10^9 \end{bmatrix} \text{ mm}^4$$

$F_{DN} = 1$

Call Program 2 to find the Stiffness Matrix of the Substitute Frame

$$K1 := \text{Find}_K(L, I_{UC}, I_{LC}, I_{beam}, H_{UC}, H_{LC}, F_{DN}) = \begin{bmatrix} 5.8969 & 1.8 & 0 & 0 \\ 1.8 & 11.2969 & 2.7 & 0 \\ 0 & 2.7 & 11.2969 & 1.8 \\ 0 & 0 & 1.8 & 5.8969 \end{bmatrix} \text{ kNm}$$

$G_k = 25 \text{ kN/m}$, and variable action, $Q_k = 10 \text{ kN/m}$, uniformly distributed along the beam.



$$G_k := 25 \frac{\text{kN}}{\text{m}}$$

$$Q_k := 10 \frac{\text{kN}}{\text{m}}$$

$$W_{max} := (1.35 \cdot G_k + 1.5 \cdot Q_k) = 48.75 \frac{\text{kN}}{\text{m}}$$

$$W_{min} := (1.35 \cdot G_k) = 33.75 \frac{\text{kN}}{\text{m}}$$

Define Load Patterns for All 4 Cases

Case I Case II Case III Case IV

$$FF := \begin{bmatrix} \begin{bmatrix} W_{max} \\ W_{min} \\ W_{max} \end{bmatrix} \begin{bmatrix} W_{min} \\ W_{max} \\ W_{min} \end{bmatrix} \begin{bmatrix} W_{max} \\ W_{max} \\ W_{min} \end{bmatrix} \begin{bmatrix} W_{min} \\ W_{max} \\ W_{max} \end{bmatrix} \end{bmatrix} = \begin{bmatrix} \begin{bmatrix} 48.75 \\ 33.75 \\ 48.75 \end{bmatrix} \begin{bmatrix} 33.75 \\ 48.75 \\ 33.75 \end{bmatrix} \begin{bmatrix} 48.75 \\ 48.75 \\ 33.75 \end{bmatrix} \begin{bmatrix} 33.75 \\ 48.75 \\ 48.75 \end{bmatrix} \end{bmatrix} \frac{kN}{m}$$

Call Program 7 to find support moments, upper & lower column moments

F_{DN} = 1

$$M_{all} := NEW_BMM \left(FF, K1, H_{UC}, H_{LC}, I_{beam}, I_{UC}, I_{LC}, FDN, L \right)$$

	Upper Col	Lower Col	
$M_{all} =$	$\begin{bmatrix} 37.2 \\ -22.2 \\ 22.2 \\ -37.2 \end{bmatrix}$	$\begin{bmatrix} 32.5 \\ -19.4 \\ 19.4 \\ -32.5 \end{bmatrix}$	Case I
$\begin{bmatrix} -69.7 & 135.6 \end{bmatrix} \begin{bmatrix} -93.9 & 93.9 \end{bmatrix} \begin{bmatrix} -135.6 & 69.7 \end{bmatrix}$	$\begin{bmatrix} 24.2 \\ -10.2 \\ 10.2 \\ -24.2 \end{bmatrix}$	$\begin{bmatrix} 21.1 \\ -8.9 \\ 8.9 \\ -21.1 \end{bmatrix}$	Case II
$\begin{bmatrix} -45.3 & 106.7 \end{bmatrix} \begin{bmatrix} -87.5 & 87.5 \end{bmatrix} \begin{bmatrix} -106.7 & 45.3 \end{bmatrix}$	$\begin{bmatrix} 35.7 \\ -17.4 \\ 12 \\ -24.7 \end{bmatrix}$	$\begin{bmatrix} 31.2 \\ -15.2 \\ 10.5 \\ -21.6 \end{bmatrix}$	Case III
$\begin{bmatrix} -66.9 & 147.6 \end{bmatrix} \begin{bmatrix} -115.1 & 79.7 \end{bmatrix} \begin{bmatrix} -102.2 & 46.3 \end{bmatrix}$	$\begin{bmatrix} 24.7 \\ -12 \\ 17.4 \\ -35.7 \end{bmatrix}$	$\begin{bmatrix} 21.6 \\ -10.5 \\ 15.2 \\ -31.2 \end{bmatrix}$	Case IV
$\begin{bmatrix} -46.3 & 102.2 \end{bmatrix} \begin{bmatrix} -79.7 & 115.1 \end{bmatrix} \begin{bmatrix} -147.6 & 66.9 \end{bmatrix}$			

kNm

Beam1

Beam 2

Beam 3

Get Support Moments

L R L R L R

$$M_{sup} := M_{all} \begin{bmatrix} 1 \dots \text{rows}(M_{all}) \end{bmatrix} \begin{bmatrix} 1 \dots \text{rows}(L) \end{bmatrix} = \begin{bmatrix} \begin{bmatrix} -69.7 & 135.6 \end{bmatrix} \begin{bmatrix} -93.9 & 93.9 \end{bmatrix} \begin{bmatrix} -135.6 & 69.7 \end{bmatrix} \\ \begin{bmatrix} -45.3 & 106.7 \end{bmatrix} \begin{bmatrix} -87.5 & 87.5 \end{bmatrix} \begin{bmatrix} -106.7 & 45.3 \end{bmatrix} \\ \begin{bmatrix} -66.9 & 147.6 \end{bmatrix} \begin{bmatrix} -115.1 & 79.7 \end{bmatrix} \begin{bmatrix} -102.2 & 46.3 \end{bmatrix} \\ \begin{bmatrix} -46.3 & 102.2 \end{bmatrix} \begin{bmatrix} -79.7 & 115.1 \end{bmatrix} \begin{bmatrix} -147.6 & 66.9 \end{bmatrix} \end{bmatrix} kNm$$

Case I
Case II
Case III
Case IV

Get Column Moments

$$M_{col} := M_{all} \begin{bmatrix} 1 \dots \text{rows}(M_{all}) \end{bmatrix} \begin{bmatrix} (\text{cols}(M_{all}) - 1) \dots \text{cols}(M_{all}) \end{bmatrix} = \begin{bmatrix} \begin{bmatrix} 37.2 \\ -22.2 \\ 22.2 \\ -37.2 \end{bmatrix} \begin{bmatrix} 32.5 \\ -19.4 \\ 19.4 \\ -32.5 \end{bmatrix} \\ \begin{bmatrix} 24.2 \\ -10.2 \\ 10.2 \\ -24.2 \end{bmatrix} \begin{bmatrix} 21.1 \\ -8.9 \\ 8.9 \\ -21.1 \end{bmatrix} \\ \begin{bmatrix} 35.7 \\ -17.4 \\ 12 \\ -24.7 \end{bmatrix} \begin{bmatrix} 31.2 \\ -15.2 \\ 10.5 \\ -21.6 \end{bmatrix} \\ \begin{bmatrix} 24.7 \\ -12 \\ 17.4 \\ -35.7 \end{bmatrix} \begin{bmatrix} 21.6 \\ -10.5 \\ 15.2 \\ -31.2 \end{bmatrix} \end{bmatrix} kNm$$

Case I
Case II
Case III
Case IV

Textbook Moment Distribution - 4 Iterations

Table 3.2 Moment distribution for the first loading case

	A			B				C				D	
	Cols. ($\sum M$)	AB		BA	Cols. ($\sum M$)	BC		CB	Cols. ($\sum M$)	CD		DC	Cols. ($\sum M$)
D.F.s Load kN	0.39	0.61	292	0.32	0.20	0.48	135	0.48	0.20	0.32	292	0.61	0.39
F.E.M.		+		-		+		-		+		-	
		146		146		45.0		45.0		146		146	
Bal.	-	-		+	+	-		-	-	-		+	+
	56.9	89.1		32.3	20.2	48.5		48.5	20.2	32.3		89.1	56.9
		+		-		-		+		+		-	
C.O.		16.2		44.6		24.2		24.2		44.6		16.2	
	-	-		+	+	-		-	-	-		+	+
Bal.	6.3	9.9		22.0	13.8	33.0		33.0	13.8	22.0		9.9	6.3
		+		-		-		+		+		-	
C.O.		11.0		5.0		16.5		16.5		5.0		11.0	
	-	-		+	+	+		-	-	-		+	+
Bal.	4.3	6.7		6.9	4.3	10.3		10.3	4.3	6.9		6.7	4.3
		+		-		-		+		+		-	
C.O.		3.4		3.4		5.2		5.2		3.4		3.4	
	-	-		+	+	+		-	-	-		+	+
Bal.	1.3	2.1		2.8	1.7	4.1		4.1	1.7	2.8		2.1	1.3
M (kN m)	-	+		-	+	+		-	-	+		-	+
	68.8	68.8		135.0	40.0	95.0		95.0	40.0	135.0		68.8	68.8

CASE 1

Moment Distribution - EXCEL										
37.16			-22.20				22.20			-37.16
0.01			-0.01				0.01			-0.01
0.01			-0.02				0.02			-0.01
0.04			-0.05				0.05			-0.04
0.10			-0.12				0.12			-0.10
0.28			-0.33				0.33			-0.28
0.71			-0.92				0.92			-0.71
2.28			-2.32				2.32			-2.28
3.35			-7.46				7.46			-3.35
30.38	A	Load Max	B			Load Min	C		Load Max	D
0.21	0.61		0.32	0.11	0.48		0.48	0.11	0.32	
0.18	-146.25		146.25	0.09	-45.00		45.00	0.09	-146.25	
26.58	89.28		-32.27	-9.61	-48.40		48.40	9.61	32.27	
	-16.13		44.64		24.20		-24.20		-44.64	
2.93	9.85		-21.94	-6.53	-32.91		32.91	6.53	21.94	
	-10.97		4.92		16.45		-16.45		-4.92	
1.99	6.70		-6.81	-2.03	-10.22		10.22	2.03	6.81	
	-3.41		3.35		5.11		-5.11		-3.35	
0.62	2.08		-2.70	-0.80	-4.04		4.04	0.80	2.70	
	-1.35		1.04		2.02		-2.02		-1.04	
0.24	0.82		-0.98	-0.29	-1.46		1.46	0.29	0.98	
	-0.49		0.41		0.73		-0.73		-0.41	
0.09	0.30		-0.36	-0.11	-0.55		0.55	0.11	0.36	
	-0.18		0.15		0.27		-0.27		-0.15	
0.03	0.11		-0.13	-0.04	-0.20		0.20	0.04	0.13	
	-0.07		0.06		0.10		-0.10		-0.06	
0.01	0.04		-0.05	-0.01	-0.07		0.07	0.01	0.05	
	-0.02		0.02		0.04		-0.04		-0.02	
0.00	0.02		-0.02	-0.01	-0.03		0.03	0.01	0.02	
	-0.01		0.01		0.01		-0.01		-0.01	
0.00	0.01		-0.01	0.00	-0.01		0.01	0.00	0.01	
	0.00		0.00		0.01		-0.01		0.00	
0.00	0.00		0.00	0.00	0.00		0.00	0.00	0.00	
32.52	-69.68		135.59	-19.43	-93.95		93.95	19.43	-135.59	